

Surface viscous stress over wind-driven waves with intermittent airflow separation

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(Received 7 February 2020; revised 10 August 2020; accepted 26 August 2020)

The small-scale physics within the first centimetres above the wavy air–sea interface are the gateway for transfers of momentum and scalars between the atmosphere and the ocean. We present an experimental investigation of the surface wind stress over laboratory wind-generated waves. Measurements were performed at the University of Delaware’s large wind-wave-current facility using a recently developed state-of-the-art wind-wave imaging system. The system was deployed at a fetch of 22.7 m, with wind speeds from 2.19 to 16.63 m s⁻¹. Airflow velocity fields were acquired using particle image velocimetry above the wind waves down to 100 μm above the surface, and wave profiles were detected using laser-induced fluorescence. The airflow intermittently separates downwind of wave crests, starting at wind speeds as low as 2.19 m s⁻¹. Such events are accompanied by a dramatic drop in tangential viscous stress past the wave’s crest, and a gradual regeneration of the viscous sublayer upon the following (downwind) crest. This contrasts with non-airflow separating waves, where the surface viscous stress drop is less significant. Airflow separation becomes increasingly dominant with increasing wind speed and wave slope ak (where a and k are peak wave amplitude and wavenumber, respectively). At the highest wind speed (16.63 m s⁻¹), airflow separation occurs over nearly 100 % of the wave crests. The total air–water momentum flux is partitioned between viscous stress and form drag at the interface. Viscous stress (respectively form drag) dominates at low (respectively high) wave slopes. Tangential viscous forcing makes a minor contribution (~3 %) to wave growth.

Key words: air/sea interactions, surface gravity waves, wind–wave interactions

1. Introduction

The dynamics in the airflow above wind-generated waves is crucial for wind–wave coupling and for the air–sea momentum flux as a whole (Janssen 1989; Komen *et al.* 1994; Belcher & Hunt 1998; Edson & Fairall 1998; Janssen 1999; Sullivan & McWilliams

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2002; Sullivan *et al.* 2008; Mueller & Veron 2009; Sullivan & McWilliams 2010; Grare, Lenain & Melville 2013a; Suzuki, Hara & Sullivan 2013; Hara & Sullivan 2015; Grare, Lenain & Melville 2018). Detailed experimental investigations of the airflow structure and wind stress above wind waves remain rare, however, largely because of the technical challenges involved with acquiring high resolution measurements very close to a rapidly moving interface (Buckley & Veron 2017). Yet, it is now accepted that surface waves in general, and airflow separation and wave breaking processes in particular, play an important role in the momentum flux between the ocean and the atmosphere (Melville 1996; Sullivan & McWilliams 2010). Thus, there is a need for detailed airflow and surface stress measurements above wind waves. This is particularly true in high wind conditions where momentum flux parametrizations still need to be improved in order to help better predict extreme weather events such as tropical storms (e.g. Edson *et al.* 2013; Veron 2015).

At moderate to high wind speeds, and/or when waves break, airflow separation is believed to have an important impact on the total air–sea momentum flux. However, our current understanding of airflow separation over wind waves is still rather poor and detailed measurements are lacking. Consequently, the effects of airflow separation on the surface stress and the range of wind speeds over which separation substantially affects the air–sea momentum flux are not well resolved.

Airflow separation began receiving increased interest over four decades ago, when Banner & Melville (1976) conducted flow-visualization studies and established on theoretical grounds derived from the earlier work of Banner & Phillips (1974), that airflow separation over a surface gravity wave occurs concurrently with breaking (see also the work of Gent & Taylor 1977). A year later, Okuda, Kawai & Toba (1977) quantified the surface stress over wind waves at relatively short fetches (2.85 m) and noted that the dominant mechanism for transferring energy from the air to the water was the high tangential stress at the interface. This was in contrast with the results of Banner & Melville (1976), who found that the normal, not tangential stress, was the main contributor to the momentum flux between air and water. Neither study discussed the instantaneous turbulent structures present when the airflow separates. In order to fill this gap, Kawai (1981, 1982) performed a qualitative flow visualization study over short wind waves and was able to detect the separated airflow. He suggested that airflow separation was caused by the inability of the airflow to curve along the steep slope of the wave crest. Later, Kawamura & Toba (1988) performed instantaneous velocity-shear measurements in the separated flow over wind waves using a pair of surface following hot-films. They explained the abrupt change in the velocity across an observed shear layer by the existence of a separation region behind the wind-wave crest. Reul, Branger & Giovanangeli (1999) pioneered the use of PIV (particle image velocimetry) in the airflow over waves and quantified airflow separation over mechanically generated breaking waves. They reported the presence of multiple coherent patches of vorticity downwind of the crest that were strongly influenced by the geometry of the breaker (see also Reul, Branger & Giovanangeli 2008). More recently, Veron, Saxena & Misra (2007) were able to identify airflow separation events over wind-generated waves. Later, Buckley & Veron (2016) and Buckley & Veron (2017) determined that separation dramatically enhances the along-wave turbulent stress and kinetic energy distributions past the crest of the average wind wave. Subsequent analysis is also presented in Buckley & Veron (2019), Husain *et al.* (2019) and Yousefi, Veron & Buckley (2020). Finally, two important laboratory studies that focused on estimating the near-surface viscous stress at the wavy air–water interface should be mentioned here. First, Banner & Peirson (1998), using underwater PIV measurements under wind-generated waves, estimated along-wave near-surface tangential stresses. Grare *et al.* (2013b) were able to estimate near-surface viscous stresses very close to the surface

by plunging a hot-wire anemometer into the wavy water. In both of these studies, however, a quantification of airflow separation effects was not possible.

Coincidentally, a number of direct numerical simulation (DNS) and large eddy simulation (LES) investigations of the airflow and airflow separation above solid wavy surfaces were initiated (Kuzan, Hanratty & Adrian 1989; Maass & Schumann 1994; Gong, Taylor & Dornbrack 1996; Angelis, Lombardi & Banerjee 1997; Henn & Sykes 1999). In particular, Maass & Schumann (1994) and Calhoun & Street (2001) studied the turbulent flow near the wavy wall and confirmed that a wavy surface substantially modifies classical wall turbulence. They also provided some insight into the detailed three-dimensional instantaneous structures in the separated region. Later, Sullivan, McWilliams & Moeng (2000) and Shen *et al.* (2003) modelled the airflow over a moving sinusoidal wall. In recent years, LES and DNS simulations have considerably improved and allowed for increasing complexity to be accounted for. In particular, multi-modal surface geometry and propagating surface waves can now be incorporated in the simulation with good success (Yang & Shen 2010; Sullivan, McWilliams & Patton 2014; Hara & Sullivan 2015; Sullivan *et al.* 2018; Wang *et al.* 2020) but available experimental data to validate these simulations are lacking.

In this paper, we present two-dimensional high resolution measurements of the airflow above wind-generated waves, within the airflow's viscous sublayer and above. We further investigate the mean and phase-averaged contributions of viscous stress to the total momentum flux across the air–water interface and the energy flux contributing to wind-wave growth. Finally, the impact of airflow separation on the average surface viscous stress is discussed.

2. Experimental set-up and methods

The experiments presented in this paper were performed in the University of Delaware's large wind-wave-current facility. The tank, specially designed for the study of air–sea interactions, is 42 m long, 1 m wide and 1.25 m high. The mean water depth was kept at 0.70 m, with an airflow space of 0.55 m. For this study, wind waves were generated by the computer-controlled recirculating wind tunnel, sketched in [figure 1\(a\)](#). In this paper, we present results for six different wind/wave conditions. Winds were generated with mean 10 m equivalent speeds of $U_{10} = 0.86, 2.19, 5.00, 9.41, 14.34$ and 16.63 m s^{-1} . The data were collected at a fetch of 22.7 m. The different experimental conditions are summarized in [table 1](#).

A complex imaging system was developed for the study of the airflow above wind waves. The experimental set-up, combining PIV with LIF techniques, allowed us to simultaneously obtain two-dimensional velocity fields in the air above the wind-generated waves, together with spatial and temporal wave properties. The apparatus and data processing techniques are described in detail in Buckley & Veron (2017). Accordingly, we will present below a brief summarized overview of the complete system.

2.1. Airflow velocity measurements

In order to obtain high resolution, two-dimensional velocity measurements in the airflow above surface wind waves, we used a custom built PIV–LIF system. For this, the airflow was seeded with 8 to 12 μm water droplets generated by a commercial fog generator and illuminated with a high intensity pulsed green laser light sheet (Nd-Yag, 200 mJ pulse⁻¹, 3–5 ns pulse duration). The light sheet was aligned in the along-wind direction and positioned in the centre line of the tank (see [figure 1](#)). Sequential image pairs of the

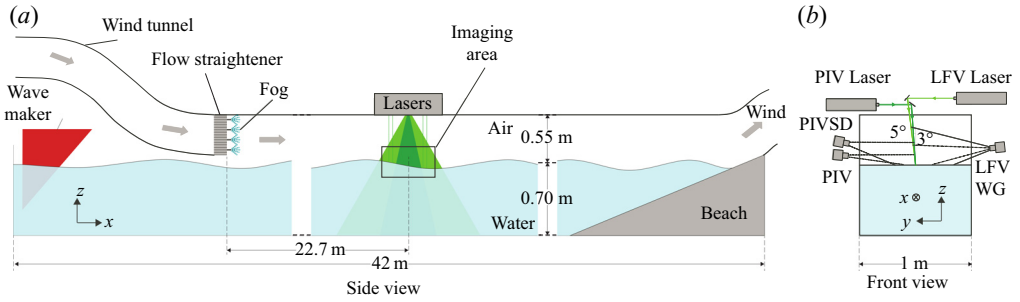


FIGURE 1. Sketch of University of Delaware’s large wind-wave tank (a). The experimental set-up is positioned at a fetch of 22.7 m. Upwind view of the imaging system (b). The optical paths of the pulsed Nd-Yag laser sheets are shown. Velocities in the airflow are measured by PIV, and wave properties are measured by laser-induced fluorescence (LIF). LIF systems include the PIVSD camera (for water surface detection on PIV images), the large field of view (LFV) laser and camera (for large field of view snapshots of wave profiles) and optical wave gauge (WG) cameras.

	U_{10} (m s ⁻¹)	u_* (cm s ⁻¹)	z_0^+	$\delta_\nu k$	kz_0 ($\times 10^{-3}$)	f_m (Hz)	c (m s ⁻¹)	k (m ⁻¹)	a (cm)	ak	c/U_{10}	c/u_*	u_d (cm s ⁻¹)
No waves	0.86	2.6	0.03	—	—	—	—	—	—	—	—	—	—
Wind	2.19	7.3	0.21	0.090	1.86	3.3	0.48	43.20	0.15	0.06	0.22	6.57	0.01
waves	5.00	16.7	0.50	0.021	1.04	2.5	0.65	23.10	0.50	0.12	0.13	3.91	2.7
	9.41	31.4	0.96	0.007	0.66	2.0	0.82	14.44	1.20	0.17	0.09	2.62	4.5
	14.34	53.8	6.45	0.003	1.88	1.8	0.97	10.47	1.96	0.20	0.07	1.80	11.2
	16.63	67.2	17.68	0.002	3.59	1.7	1.04	9.05	2.29	0.21	0.06	1.55	13.9

TABLE 1. Summary of the experimental conditions. Measured, apparent peak wave frequencies f_m were obtained from the optical wave gauge frequency spectra. The wave speed c denotes the intrinsic phase speed and was extracted using a cross-spectral analysis of two consecutive optical wave gauge signals which included the wind-induced drift u_d . The wave amplitude a is estimated using $a = \sqrt{2}a_{rms}$, where a_{rms} is the root-mean-squared amplitude measured with optical wave gauges. The wall-normalized roughness length (or roughness Reynolds number, Donelan 1998) is given by $z_0^+ = u_* z_0 / \nu$ (where z_0 is the roughness length and ν the air kinematic viscosity). Both u_* and z_0 are estimated from the mean velocity profiles assuming classical log behaviour outside the wave boundary layer. U_{10} is the mean air velocity extrapolated from the measurements to $z = 10$ m, obtained by fitting the upper logarithmic part of each mean velocity profile. $\delta_\nu k$ is the thickness of the viscous sublayer scaled by wavenumber k , with $\delta_\nu = 10\nu/u_*$ (e.g. Phillips 1977, p. 128).

seeded flow were obtained with two side-by-side CCD cameras (JAI, 2048 × 2048 pixel). The adjacent PIV images were merged together in order to obtain a high resolution (47 μm pixel⁻¹) 18.7 cm × 9.7 cm PIV image. The resulting PIV image pairs were then processed using conventional cross-correlation techniques (Raffel *et al.* 2007) performed using an in-house iterative multi-grid software. Two-dimensional velocity maps were thus obtained with a velocity vector measurement every 188 μm² over the 18.7 cm × 9.7 cm imaging footprint. The PIV cameras operated at 14.4 frames/second yielding PIV velocity maps at a 7.2 Hz rate. Low laser light reflection from the air–water interface rendered

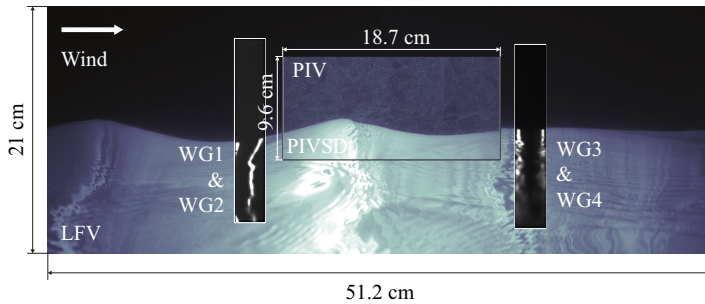


FIGURE 2. A composite image showing the air-side portion of merged raw PIV images, plotted above the water-side portion of the PIVSD image used for surface detection. The background image is the LFV LIF. WG fields of view are also displayed, up and downwind of the PIV field of view. This sextuplet of instantaneous PIV–LIF images was acquired during the experiment with $U_{10} = 5.00 \text{ m s}^{-1}$.

precise automated surface detection difficult on the raw PIV images. To address this issue, high resolution ($100 \mu\text{m pixel}^{-1}$) LIF images (PIV surface detection images, noted ‘PIVSD’) of the wave surface profiles within the PIV field of view were acquired simultaneously with the PIV images. This was achieved by adding a liquid solution of Rhodamine 6G dye to the water at a concentration of $8 \times 10^{-6} \text{ g L}^{-1}$. The fluorescent dye was then excited by the PIV laser and images of the surface were collected with a CCD camera identical to those used for the PIV. An optical filter was also used to remove green laser light reflections (and potential contamination from light scattered by the fog particles), and isolate the light emitted by the fluorescent dye. This yielded sharp, crisp images of the interface which was then automatically detected using an in-house, gradient-based edge-detection image processing software (as pioneered by Duncan (1981), Duncan *et al.* (1999), for the study of mechanically generated waves). In turn, being able to properly locate the air–water interface on the PIV images allowed us to perform velocity measurements very close to the interface. In the end, airflow velocities were consistently measured as close as $100 \mu\text{m}$ above the air–water interface. A sample merged PIV–PIVSD image is plotted in figure 2.

2.2. Wind-wave measurements

The LIF technique employed to precisely locate the air–water interface within the PIV frame was also used to measure two additional types of wind wave data: along-channel spatial surface profiles with high spatial resolution, and single point high frequency wave height measurements.

Large along-channel spatial profiles of the wavy surface were obtained by LIF, using a third CCD camera (JAI, 2048×2048 pixel, noted ‘LFV’ hereafter), that was focused on the intersection with the surface of a long green laser sheet, generated by a second pulsed Nd-Yag laser ($120 \text{ mJ pulse}^{-1}$, 3–5 ns pulse duration). The resulting LFV images provided measurements of the along-channel surface elevation in the centre line of the channel over a length of 51.2 cm ($0.25 \text{ mm pixel}^{-1}$ resolution), and at a rate of 7.2 Hz. The LFV field of view was positioned in the same along-channel plane as the PIV images, and extended 16.7 cm upwind and 15.8 cm downwind of the PIV field of view (figure 2).

Finally, four single point optical wave gauges, provided time series of the water height 2.8 and 1.4 cm upwind and 2.7 and 4.2 cm downwind of the PIV airflow velocity measurements. The system consisted of two CCD cameras (JAI, 300×1600 pixel), which

imaged the intersection with the surface of pairs of 200 mW continuous green laser beams. The resulting LIF images provided measurements of the water height with a resolution of $65 \mu\text{m pixel}^{-1}$, and at a frequency of 93.6 Hz (figure 2). Peak apparent wave frequencies f_m were obtained from the optical wave gauge frequency spectra. Here, the peak frequencies are Doppler shifted by the wind-induced drift

$$\omega_m = \sigma + ku_d = k(c + u_d), \quad (2.1)$$

where $\omega_m = 2\pi f_m$ is the apparent (measured) angular frequency, c and σ are the intrinsic phase speed and angular frequency respectively and k is the wavenumber (e.g. Phillips 1977, p. 24). Peak wave speeds $c_m = c + u_d$ were directly estimated using a cross-spectral analysis on two adjacent optical wave gauge signals. The wavenumber k was then readily obtained using (2.1). Next, the intrinsic frequency σ and wave phase speed c were estimated using the linear dispersion relationship for deep water gravity waves $\sigma = ck = \sqrt{gk + \Gamma k^3}$ where Γ is the surface tension. The wind drift velocity u_d is given by $u_d = (\omega_m - \sigma)/k$. Note that the Doppler shifting results from a drift current that acts not only on the surface characteristics of the waves, but also on the underwater wave orbital motions. This means that u_d is an integrated measure of the wind-induced shear currents down to a depth of $O(0.1\lambda)$ (e.g. Stewart & Joy 1974), rather than a solely surface wind drift velocity u_s (see also Smeltzer *et al.* 2019). The peak wave amplitude a was estimated using $a = \sqrt{2}a_{rms}$, where a_{rms} is the root-mean-squared amplitude measured with optical wave gauges (see table 1).

2.3. Experimental procedure

All cameras and pulsed lasers were synchronized using triggers from a single timing computer equipped with National Instruments digital and analogue signal generation software and hardware (see Buckley & Veron (2017), for additional details). The wind blower was controlled by an analogue signal, also generated by the timing computer. Each wind-wave experiment followed a fully automated, repeatable procedure. At first, the wind was slowly increased to its target steady value. After the wave field had sufficiently developed and reached a fetch-limited equilibrium state, the fog generator started and the system acquired simultaneously PIV data, LIF PIV surface detection data, LIF large field of view data and LIF single point wave height data. During each experiment, the inside of the tank windows were dried using window wipers every 30 s, and for a period of 3 s. The images altered by the presence of the wiper were later systematically removed from the dataset. Experiments were performed for durations varying from 4.5 to 14 min, depending on the wind-wave conditions. These durations were chosen based on the estimated dominant wavelength and wave speed for each experiment, with the objective of sampling approximately 2000 distinct waves per experiment, regardless of the wind conditions. Overall, the complete dataset consists of nearly 2 million images. Acquired images were transferred to hard drive storage arrays from the frame grabbers, then processed using Matlab (MathWorks).

3. Results

3.1. High resolution airflow velocity fields

Representative examples of instantaneous horizontal velocity fields obtained in the airflow are shown in figure 3 for all six wind speeds studied. Here, u , the horizontal velocity components of the velocity vectors \mathbf{u} are plotted above the corresponding LIF image used to detect the position of the interface. When $U_{10} = 0.86 \text{ m s}^{-1}$ (figure 3a), no apparent

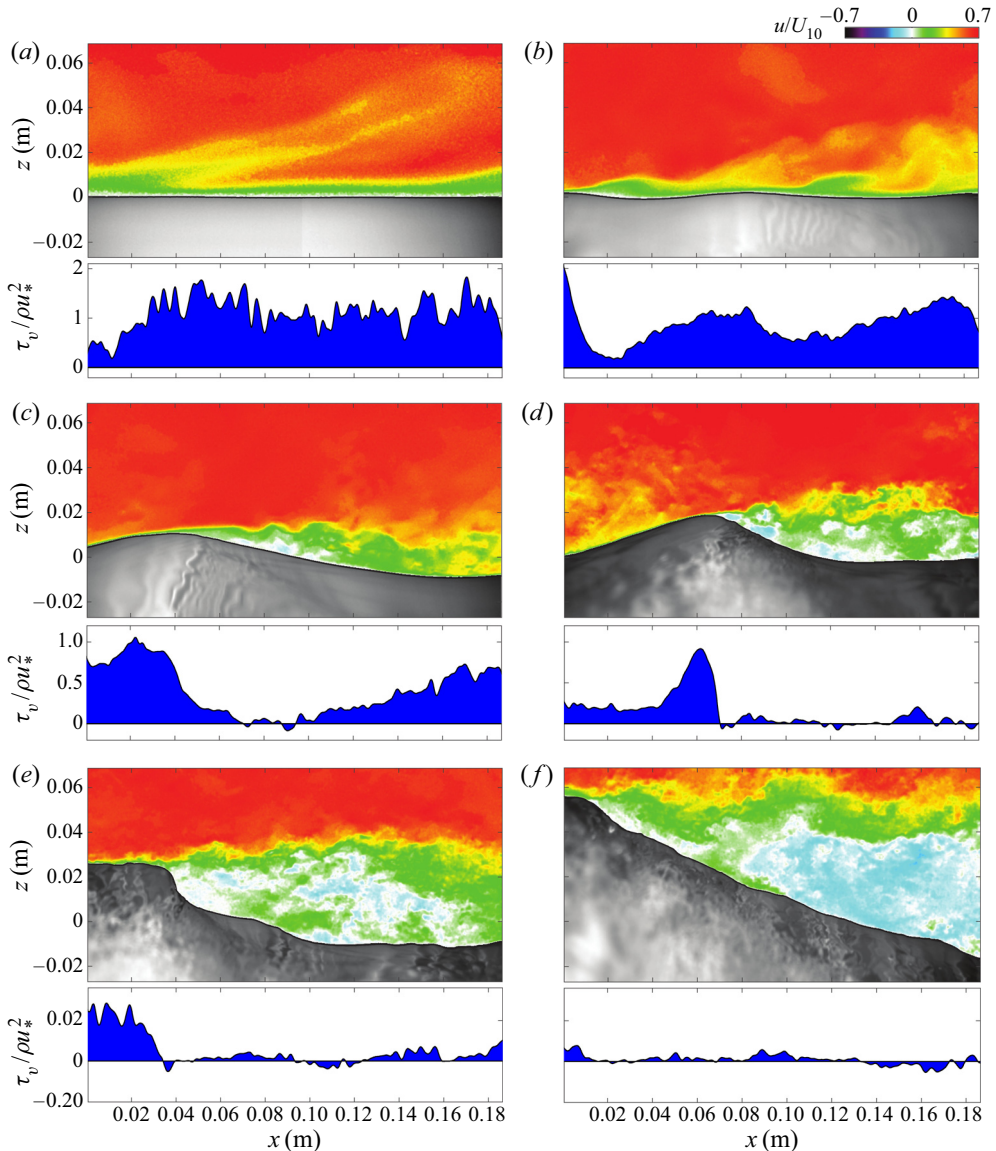


FIGURE 3. Instantaneous velocities and surface viscous stress, for six different wind-wave conditions, $U_{10} = 0.86 \text{ m s}^{-1}$ (a), 2.19 m s^{-1} (b), 5.00 m s^{-1} (c), 9.41 m s^{-1} (d), 14.34 m s^{-1} (e) and 16.63 m s^{-1} (f). Estimates of the surface viscous stresses are shown in the panels below the wave shapes and are averages from 284 to 664 μm above the water surface, normalized by the total mean wind stress.

waves are generated. In fact, in this case the instantaneous airflow velocity field shows structures that can be interpreted as sweeps of high velocity fluid toward the surface and ejection of low velocity fluid into the bulk fluid. These ejection and sweep events are typical of turbulent boundary layers over solid flat boundaries (Kline *et al.* 1967; Adrian 2007).

At $U_{10} = 2.19 \text{ m s}^{-1}$ (figure 3b), short surface waves are present. Here, the ejections and sweeps, as well as a local thinning and thickening of the near-surface boundary layer,

appear to be coherent with surface waves. The alternating thinning and thickening of the turbulent boundary layer above waves (without airflow separation) was in fact suggested by Belcher & Hunt (1993), who introduced it under the term ‘non-separated sheltering’. For a further analysis of this phenomenon, the reader is referred to the recent study of Buckley & Veron (2019), where the wave-coherent quadrants of the turbulent Reynolds stress tensor are examined, as well as to the work of Sullivan & McWilliams (2010), who have shown the importance of ejections and sweeps for downward turbulent momentum flux across the ocean surface in young wind-wave conditions.

At wind speeds $U_{10} = 5.00, 9.41, 14.34$ and 16.63 m s^{-1} (figure 3c–f), the turbulent boundary layer is unequivocally modified by the presence of the surface waves. Above the crest of the waves, the airflow moves fast throughout the entire height of the sampled air column, and u decreases substantially only very near the surface within the viscous layer. This implies strong shear and shear stress near the surface. In turn, this is consistent with favourable pressure gradient conditions similar to what would occur in an aerodynamic flow over a fixed obstacle (Baskaran, Smits & Joubert 1987; Simpson 1989). Just past the wave crests, the air boundary layer appears to detach from the water surface, and the near-surface streamwise velocity drops dramatically. We observe a region of negative velocity on the lee side of the surface waves indicating that the crest of the waves are completely sheltering this region from the bulk flow.

In figure 4, we provide the averaged velocity profile for each of the six wind conditions. At the lowest wind speed ($U_{10} = 0.86 \text{ m s}^{-1}$), where no waves were observed, the averaged velocity profile conforms to the log–linear ‘law of the wall’. At this low wind speed, we are able to resolve the viscous sublayer with a resolution of approximately $\zeta^+ = 0.33$ where $\zeta^+ = \zeta u_* / \nu$ is the wall coordinate, i.e. the distance from the boundary ζ normalized using the friction velocity u_* , and the kinematic viscosity of the air ν . Because of the presence of the wavy surface, average wind speed profiles are necessarily obtained using ζ as the vertical distance from the surface (see Buckley & Veron 2017). As the wind speed increases, the mean profiles depart from the law of the wall, as the presence of waves increasingly modifies the airflow. A widespread approach to assess the influence of waves on the mean wind profile is through the aerodynamic roughness. Kitaigorodskii & Donelan (1984) suggest that the ocean surface be aerodynamically smooth when $z_o^+ \sim 0.1$, transitional when $0.1 < z_o^+ < 2.2$, and fully rough when $z_o^+ > 2.2$ (see also Donelan 1998). In this classification, our low wind no-wave and low wind-wave cases ($U_{10} = 0.86 \text{ m s}^{-1}$, $U_{10} = 2.19$), where the air velocities show features reminiscent of the airflow over a smooth plate (see again figure 3a,b), are just below and just above the aerodynamically smooth regime, respectively; the aerodynamic conditions with $U_{10} = 5.00 \text{ m s}^{-1}$, $U_{10} = 9.41 \text{ m s}^{-1}$, figure 3(c,d) are transitional; and the two highest wind speed cases ($U_{10} = 14.34 \text{ m s}^{-1}$, $U_{10} = 16.63 \text{ m s}^{-1}$), with their dramatic airflow separation events, are in the fully rough regime (figure 3e,f).

Finally, it is important to note here that velocity measurements alone are not Galilean invariant and thus cannot unambiguously identify separation (Simpson 1989). Data products derived from the velocity gradient tensor $\nabla \mathbf{u}$ such as vorticity or viscous stresses (see below) are generally more robust indicators of airflow separation (Simpson 1989; Wu 2000).

3.2. Surface tangential viscous stress

The surface tangential viscous stress, or skin friction drag, (see Schlichting & Gersten 2000) is estimated with

$$\tau_v = (\boldsymbol{\tau n}) \cdot \mathbf{t}, \quad (3.1)$$

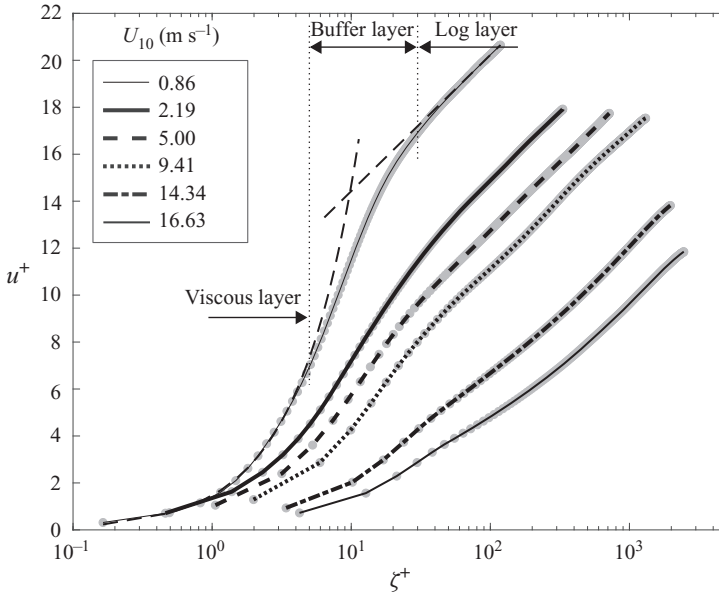


FIGURE 4. Mean wind profiles in wall coordinates for all six wind conditions. Best fits of the respective linear and log parts of the profile are shown in dashed lines for the case $U_{10} = 0.86 \text{ m s}^{-1}$.

where $\boldsymbol{\tau} = \mu(\nabla\mathbf{u} + \nabla\mathbf{u}^T)$ is the air-side viscous stress tensor (and estimated at the interface), μ is the air dynamic viscosity and \mathbf{n} and \mathbf{t} are unit vectors that are respectively normal and tangent to the local instantaneous water surface (see Buckley & Veron 2017). A diagram of the problem’s geometry is shown in figure 5. Here, we measured velocities in the vertical plane aligned with the direction of wave propagation, $\mathbf{u} = (u, w)$, so

$$\tau_v = \mu \frac{1 - \epsilon^2}{1 + \epsilon^2} \left(\left(\frac{\partial u}{\partial z} + \frac{\partial w}{\partial x} \right) + 2 \frac{\epsilon}{1 - \epsilon^2} \left(\frac{\partial w}{\partial z} - \frac{\partial u}{\partial x} \right) \right), \quad (3.2)$$

where η is the surface elevation and $\epsilon = \partial\eta/\partial x$ is the surface slope. To first order in ϵ (see also Longuet-Higgins 1969a),

$$\tau_v = \mu \left(\left(\frac{\partial u}{\partial z} + \frac{\partial w}{\partial x} \right) + 2\epsilon \left(\frac{\partial w}{\partial z} - \frac{\partial u}{\partial x} \right) \right). \quad (3.3)$$

The bottom panels below the instantaneous horizontal velocity fields (figure 3) show the along-wave (tangential) surface viscous stress measurements from (3.2), taken within the airflow’s viscous sublayer (averaged from 284 to 664 μm above the air–water interface). The measurements are normalized by the total stress, which we estimate with ρu_*^2 where ρ is the mass density of the air, and u_* is the air-side friction velocity obtained from the classical log fit to the mean wind speed profile taken outside of the wave boundary layer (Buckley & Veron 2017).

At $U_{10} = 2.19 \text{ m s}^{-1}$ (figure 3b), the nascent short surface waves appear to modulate the surface viscous stress through the sweeps and ejections events and the associated alternating thinning and thickening the viscous sublayer. At wind speeds $U_{10} = 5.00 \text{ m s}^{-1}$ and higher (figure 3c–f), when the airflow separates, the viscous stress generally peaks at or near the crest, then dramatically drops and becomes nearly null just past the crest

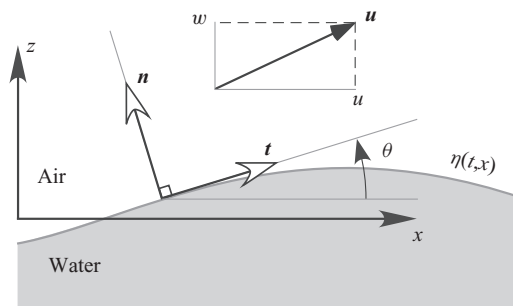


FIGURE 5. Schematic of the surface following coordinate system, at the wavy water surface. We also note $\epsilon = \partial\eta/\partial x$ and $\theta = \text{atan}(\epsilon)$.

indicating separation of the airflow. In cases where the near-surface flow reverses, the surface viscous stress becomes negative. In this study, we take a null or negative surface viscous stress as an indication that airflow separation occurs.

We note here that previous viscous stress estimates consider only vertical shear $\partial u/\partial z$ (e.g. Reul *et al.* 1999; Grare *et al.* 2013*b*). However, while $\partial w/\partial x$, $\partial w/\partial z$, and $\partial u/\partial x$ are expected to be vanishingly small on average, occasional airflow separation events such as those shown in figure 3 clearly induce local, instantaneous near-surface streamwise and vertical variability in the flow field resulting in local intermittency in the surface viscous stress. In particular, $\partial u/\partial x$ may be significant, even when multiplied by the wave slope (3.3). Note that, at the interface, w scales with the wave orbital velocity, and $\partial w/\partial x$ and $\partial w/\partial z$ thus also scale with the wave slope.

In figure 6, we show two examples of waves during the same experimental run (with $U_{10} = 5.00 \text{ m s}^{-1}$), one of which causes airflow separation (figure 6*a*), while the other causes non-separated sheltering (figure 6*b*). Over the airflow separating wave (one can see the reversal in the near-surface airflow), the viscous stress τ_v (third row in figure 6) slightly exceeds the mean total stress ρu_*^2 at the wave crest, drops down to a negative then a near-zero value downwind of the crest, and slowly increases back up to $\sim 50\%$ of the total stress as the viscous shear layer is gradually regenerated at the surface on the windward side of the next wave. Over the non-separating wave, the viscous stress is greatest just before the crest where it also slightly exceeds the total stress, and decreases past the crest to $\sim 15\%$ of the total stress. It neither approaches zero nor becomes negative; it also slowly increases again on the windward face of the next downwind wave. Note that the along-wave variations of the viscous stress broadly emulate those of the local surface slope $\partial\eta/\partial x$ (plotted in the second row in figure 6), with an abrupt (respectively less abrupt) drop near the crest in the airflow separating (respectively non-separating) case, and a gradual increase farther downwind. For completeness, we also show in figure 6 all the terms in (3.3). Clearly, the viscous tangential stress is dominated by $\partial u/\partial z$. However, as suggested above, the along-wave variability of the airflow, even multiplied by the local wave slope, $\epsilon(\partial u/\partial x)$, may be of significance, at least locally, in cases where the airflow separates. Terms involving the vertical velocity w are negligible. Although described using single events shown on figures 3 and 6 as an example, the characteristics of the airflow in separated and non-separated cases hold robustly for all the observed cases.

3.3. Phase-averaged along-wave viscous stresses

The wave-phase-averaged near-surface tangential viscous stresses $\langle \tau_v \rangle$ are plotted in figure 7. Prior to averaging, the near-surface viscous stress estimates are obtained by

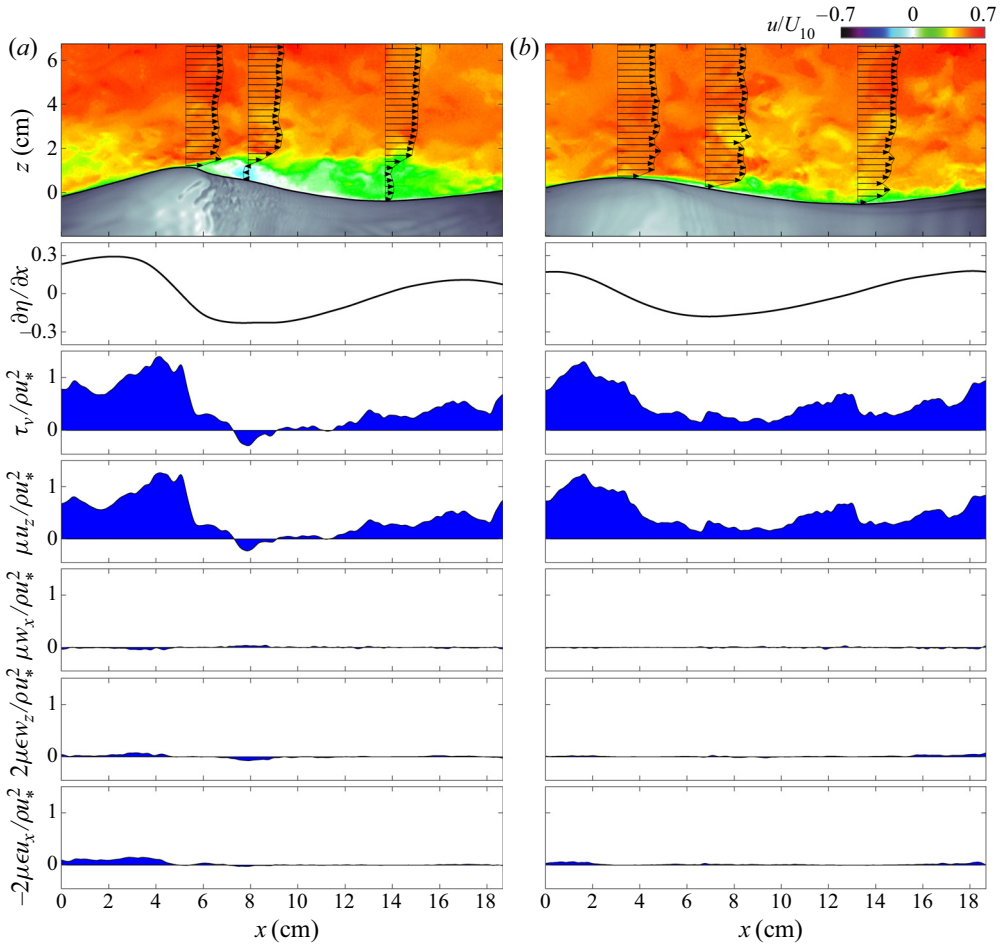


FIGURE 6. Instantaneous velocities and surface viscous stress for an airflow separating wave (a) and a non-separated sheltering wave (b), with $U_{10} = 5.00 \text{ m s}^{-1}$. Below the velocities, we show the local wave slope $\epsilon = \partial\eta/\partial x$. The panels below each wave show the terms contributing to the local tangential viscous surface stress (3.3).

averaging stress measurements between 284 and 664 μm above the air–water interface. The wave-phase-averaging procedure is described in more detail in Buckley & Veron (2017). Briefly, the wave-phase-averaged velocity data are bin averaged according to the local phase of the peak surface wave (with wavenumber k given in table 1). First, we observe that the mean (normalized) along-wave viscous stresses are, at all phases, a non-zero fraction of the total stress ρu_*^2 . This fraction decreases with increasing wind velocity. The minimum mean viscous stress to total stress ratio for this study is 0.04, on the leeward face of wind waves with $U_{10} = 16.63 \text{ m s}^{-1}$. The maximum is approximately 1.1 which occurs just upwind of the crests of wind waves with $U_{10} = 2.19 \text{ m s}^{-1}$. This means that at high wind speeds, the viscous stress is a negligible fraction of the total stress, even very close to the surface ($\zeta \sim 300 \mu\text{m}$), while at low wind speeds, the surface viscous stress supports the majority of the total wind stress. The general pattern of the stress variations for all wind-wave conditions is a minimum at the middle of the leeward side of waves ($\phi \sim \pi/2$), and a peak surface stress location always upwind of wave crests.

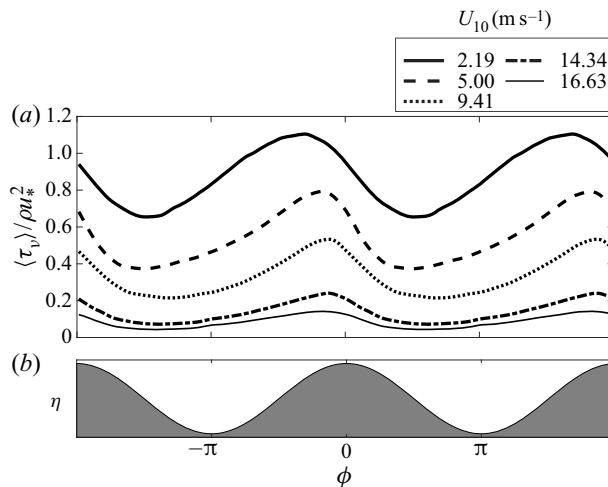


FIGURE 7. Phase-averaged viscous stress $\langle \tau_v \rangle$, taken within the viscous sublayer, and normalized by ρu_*^2 (a). A sketch of a mean wave profile is provided (b) to help visualize the wave phase. Here, the wind is blowing from left to right.

This maximum moves downwind with increasing wind speed. This is probably related to the nature and frequency of airflow sheltering events (separated or not), which, in turn depend upon wave slope and wave age (see also § 4.2 below).

We should also note that $\langle \tau_v \rangle$ presents along-wave asymmetry (or ‘vertical asymmetry’, term used to describe the geometrical properties of waves by Bonmarin (1989) for example): the along-wave variation of the phase-averaged stress is always relatively gentle upwind of the stress maximum; past the peak, the viscous stress drops more dramatically. This asymmetry is probably caused by sheltering events past wave crests.

Finally, we note that our average along-wave stress values are approximately half of the extrapolated values found by Banner & Peirson (1998) (see also figure 9b below, for a comparative study of mean viscous stress measurements). However, their study was done at a shorter fetch, with shorter, steeper, slower waves. In addition, their mean values are computed from a small number of relatively scattered measurements. Here, the mean value in each of the 144 phase bins was computed from at least 12 000 surface tangential stress measurements (and up to 48 000 at the highest wind speeds).

Next, we examine the partitioning of the mean surface wind stress (average across all wave phases) between viscous and form stress contributions, this for different wind-wave conditions.

3.4. Momentum flux: mean viscous stress and form drag

We now examine the mean surface stress where the mean is defined as an ensemble average over all measurements made along the surface, or equivalently, the average over all wave phases. We denote the ensemble mean with an overbar. For example, the mean surface viscous stress is $\overline{\tau_v}$. In other words, the phase-average viscous stress decomposes into an ensemble mean and a wave-phase-coherent variation $\tilde{\tau}_v$ which has a zero mean

$$\langle \tau_v \rangle = \overline{\tau_v} + \tilde{\tau}_v. \quad (3.4)$$

In figure 8, we compare our measurements of the mean interfacial viscous stress with those of Banner & Peirson (1998) and Grare *et al.* (2013b). We present the result as the

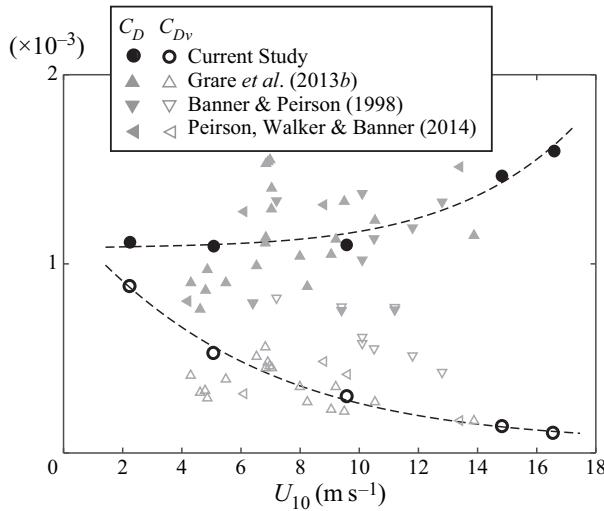


FIGURE 8. Mean total drag coefficient (C_D – solid symbols) and mean viscous drag coefficient (C_{Dv} – open symbols) as a function of 10-m wind speed U_{10} (black symbols). Drag coefficient definitions are provided in (3.6) and (3.5). For comparison, we also show data extracted from Grare *et al.* (2013*b*), Banner & Peirson (1998) and Peirson, Walker & Banner (2014) (grey symbols).

mean surface viscous drag C_{Dv} (i.e. the viscous stress scale with the 10-m wind speed) and the total surface drag C_D coefficients

$$C_{Dv} = \frac{\overline{\tau_v}}{\rho U_{10}^2}, \tag{3.5}$$

$$C_D = \frac{u_*^2}{U_{10}^2}. \tag{3.6}$$

We note that our estimates of the total drag are in good general agreement with previous measurements; although drag coefficients plotted as a function of wind speed alone generally exhibit a large scatter since wind speed is a parameter that does not necessarily capture the complexity of the air–sea interface in a given set of wind and wave conditions. In fact, the air–sea drag coefficient is known to also depend on the wave ages and wave slope (Komen *et al.* 1994). Nonetheless, in accord with numerous previous measurements, we observe a nearly constant drag at wind speeds of $U_{10} < 10 \text{ m s}^{-1}$, and an increase of the total drag with increasing wind speed when $U_{10} > 10 \text{ m s}^{-1}$. The viscous drag on the other hand, decreases with increasing wind speed, as was also observed by Banner & Peirson (1998) from underwater measurements, and by Grare *et al.* (2013*b*) from air-side measurements. Our viscous drag measurements agree well with the results obtained by Grare *et al.* (2013*b*), and are approximately 40 % lower than those of Banner & Peirson (1998). We note here that each stress measurement presented in figure 8 is obtained by averaging over 2 million PIV stress measurements taken in the airflow viscous layer near the air–water interface. We find that the viscous stress in these low to moderate wind speeds represents a non-negligible contribution to the total momentum flux which is in agreement with conclusions by Banner & Peirson (1998) and later Grare *et al.* (2013*b*).

At the interface where turbulence is suppressed, the mean horizontal stress, to first order in the wave slope, is the sum of the mean viscous stress $\overline{\tau_v}$ (from (3.3)), and the momentum

flux resulting from the pressure acting on the wavy surface, i.e. the form drag $\overline{\tau}_f = \overline{p\epsilon}$. In other words, the mean total horizontal stress at the interface naturally decomposes into viscous and pressure components. Combined, they support the total mean surface stress which can be inferred from measurements performed far from the interface at heights where both viscous and wave effects are negligible and where the mean total stress arises from the turbulence alone (where it can be scaled by a friction velocity and expressed as ρu_*^2).

In the wind-wave tank, at the measurement fetch of 22.7 m, the horizontal pressure gradient which drives the flow leads to a ‘total’ wind stress which decreases linearly with increasing height above the surface (Uz *et al.* 2002; Zavadsky & Shemer 2012; Husain *et al.* 2019). For this study, we have examined the profiles of the turbulent Reynolds stress $\rho u'w'$ outside the wave boundary layer (the primes denote turbulent velocities obtained after extracting the mean and wave-coherent components from the measured instantaneous velocity field Buckley & Veron 2017). Over the PIV measurement domain, the Reynolds stress profiles indeed exhibit a weak linear trend, however, extrapolating these profiles to the $z = 0$ does not yield a surface stress that is substantially different from ρu_*^2 . Therefore, for the purpose of this analysis, we approximate the surface stress by ρu_*^2 . Thus, at the air-water interface, the sum of the mean tangential viscous stress and the form drag is

$$\begin{aligned}\rho u_*^2 &= \overline{\tau}_v + \overline{p\epsilon}, \\ &= \overline{\tau}_v + \overline{\tau}_f.\end{aligned}\tag{3.7}$$

As noted above, both the wave slope ak and wave age c/u_* influence the air sea drag, in part because of airflow separation (we show preliminary quantitative estimates in figure 16), and the partitioning between viscous and pressure drag is thus expected to also depend on both ak and wave age c/u_* . However, in these constant fetch laboratory experiments, ak and c/u_* are tightly correlated (see table 1). Therefore, in the remainder of the paper, we will narrow our analysis to the wave slope dependence, except when wave age is a convenient parameter in order to compare our results with available data.

Using the direct measurements of u_* and $\overline{\tau}_v$, we can directly estimate $\overline{\tau}_f$ from (3.7). Figure 9(a), shows mean form drag contributions to the total stress, as a function of wave slope ak . Results from a number of other studies (Mastenbroek *et al.* 1996; Grare 2009; Peirson *et al.* 2014) are also shown. Our form drag estimates show an increase with increasing slope, and fall well within the estimates from others. The counterpart of the form drag, the viscous stress, is plotted in figure 9(b). The increase (respectively decrease) in form (respectively viscous) drag with increasing wave slope show that as waves steepen (under the action of increasing wind speeds), viscosity effects become less significant, and form drag becomes a large contributor to the total surface stress.

3.5. Surface energy flux: viscous and form stress contributions

Concurrently with the surface momentum flux, there is also an energy flux through the surface, part of which creates, sustains, or damps the surface waves. In fact, the evolution of the wave field, in deep water and without currents, can be described by a conservation equation for the wave energy spectral density (Komen *et al.* 1994)

$$\rho_w g \left(\frac{\partial \Phi(\mathbf{k})}{\partial t} + \mathbf{c}_g \cdot \nabla \Phi(\mathbf{k}) \right) = S_{in} + T_{nl} - D,\tag{3.8}$$

where ρ_w is the mass density of water, g is gravity, $\Phi(\mathbf{k})$ is the spectrum of the surface elevation and \mathbf{c}_g the surface wave group velocity. In the conservation equation above, S_{in} is

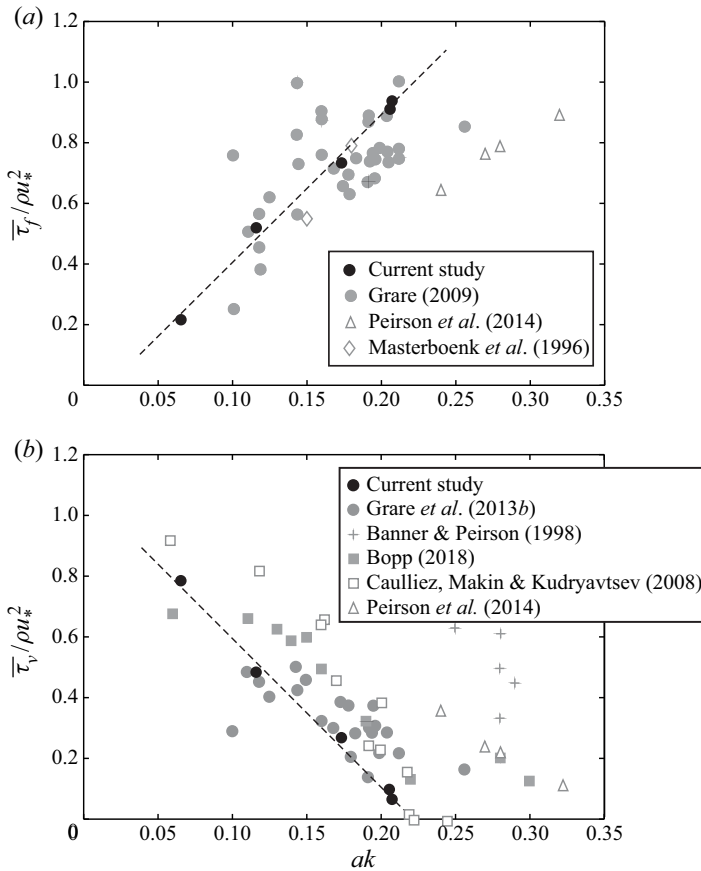


FIGURE 9. (a) Mean form drag $\overline{\tau}_f$ normalized by the total stress, as a function of wave slope (black symbols). An expression for $\overline{\tau}_f$ is provided in (3.7). Results from other published laboratory studies are also shown (Mastenbroek *et al.* 1996; Grare 2009; Grare *et al.* 2013b; Peirson *et al.* 2014). Note that extreme values of $\overline{\tau}_f / \rho u_*^2$ (greater than 1.5 and smaller than 0) reported by Grare *et al.* (2013b), were omitted. (b) Mean viscous stress normalized over total stress, as a function of wave slope, plotted alongside measurements from Banner & Peirson (1998), Caulliez, Makin & Kudryavtsev (2008), Grare *et al.* (2013b) and Bopp (2018).

a spectral wind input term, T_{nl} is an energy transport term from nonlinear wave–wave interactions, and D is a dissipation term from breaking and/or viscous effects.

While (3.8) describes the evolution of the surface waves, it is clear that the total momentum and energy fluxes across the air–water interface contribute to both waves and surface currents. Noting $\boldsymbol{\tau}_s$ the local wind stress vector on the surface, then, per unit area, the mean rate of work on the interface is

$$S_0 = \overline{\boldsymbol{\tau}_s \cdot \mathbf{u}_s}, \tag{3.9}$$

where \mathbf{u}_s is the surface velocity vector. Assuming linear spectral decomposition of the wave field, we choose to focus on the peak dominant wave and isolate the data products coherent with the peak wave component using the wave-phase-averaging procedure. For example, we note that the vertical component of the surface velocity w_s , coherent with the surface wave component of frequency f , is, to first order in ak , the surface wave vertical

orbital velocity w_o . In other words $\tilde{w}_s = w_o$ which we estimated from linear wave theory. Thus, to leading order,

$$S_0 = \overline{\tau_v u_s} - \overline{p w_o}, \quad (3.10)$$

where the viscous and pressure terms contained in τ_s appear respectively as correlations with the horizontal and vertical components of the surface velocity. Also, the vertical component of the orbital velocity can be approximated using the peak wave phase velocity c as

$$\overline{p w_o} \sim -\overline{p \epsilon c} \quad (3.11)$$

which leads to

$$\begin{aligned} S_0 &= \overline{\tau_v u_s} + \overline{\tau_f c}, \\ &= S_v + S_f, \end{aligned} \quad (3.12)$$

where $S_v = \overline{\tau_v u_s}$ and $S_f = \overline{\tau_f c}$ represent the wind input terms from the viscous and pressure forces respectively.

The surface viscous stress and horizontal surface velocity can likewise be decomposed into mean and wave-coherent components, leading to

$$S_v = \bar{\tau}_v \bar{u}_s + \widetilde{\tau}_v \widetilde{u}_s. \quad (3.13)$$

Therefore, S_v contains wind energy flux to both surface currents, $S_{vc} = \bar{\tau}_v \bar{u}_s$, and surface waves $S_{vw} = \widetilde{\tau}_v \widetilde{u}_s$. In (3.13), it is clear that the mean surface viscous stress contributes to generating mean surface drift currents while the wave-coherent surface viscous stress contributes to wave growth insofar as there is an accompanying (co-variant) wave-coherent horizontal velocity.

Thus, in order to fully estimate S_v we need to estimate mean and wave-phase-coherent components of the surface drift velocity u_s . This was accomplished by extrapolating to the interface the air-side velocity measurements obtained in the linear viscous boundary. The results were then phase averaged which allowed for the surface velocity to be further decomposed into a wave-phase-coherent and a mean component. Grare *et al.* (2013b) noted that they were not able to recover surface drift from their air-side velocity measurements and instead assumed that viscous forces did not contribute to wave growth. Here, we find that the extrapolation for air-side velocity measurements yields reliable surface estimates because our air-side airflow measurements closest to the surface are systematically within the linear viscous sublayer. We find that mean surface drifts estimates give $\bar{u}_s/u_* = 0.9 \pm 0.16$ and $\bar{u}_s/U_{10} = 3.2\% \pm 0.7\%$ which are consistent with canonical values. Also, we find that the wave-phase-coherent velocities at the interface show a phase relationship with the surface elevation that is similar to that of the surface stress. This is consistent with the observations of Veron *et al.* (2007) who also observed a one-to-one correspondence between near-surface velocity and viscous stress.

Figure 10(a) shows the wind input terms from viscosity viscous S_v and pressure S_f , normalized by the total wind input S_0 . With increasing wave slope (and thus wind speed in these fetch-limited laboratory experiments), as expected, the relative fraction of the energy imparted to the interface from frictional effect diminishes. At low wind speed and small wave slope, $ak \approx 0.1$ for example, $S_v \approx 0.35S_0$ and $S_f \approx 0.65S_0$ indicating that viscous forces do indeed provide a significant fraction of the total energy to the surface. However, as shown in figure 10(b), in these low wind speed conditions, the vast majority of the viscous input term contributes to the surface currents (wind drift) rather than the waves. The fraction of S_v which contributes to wave growth, S_{vw} , increases to approximately 0.2–0.25 for $ak \approx 0.25$ –0.3 (in fact, $S_{vw}/S_v \sim ak$), but S_v is a small fraction of the total

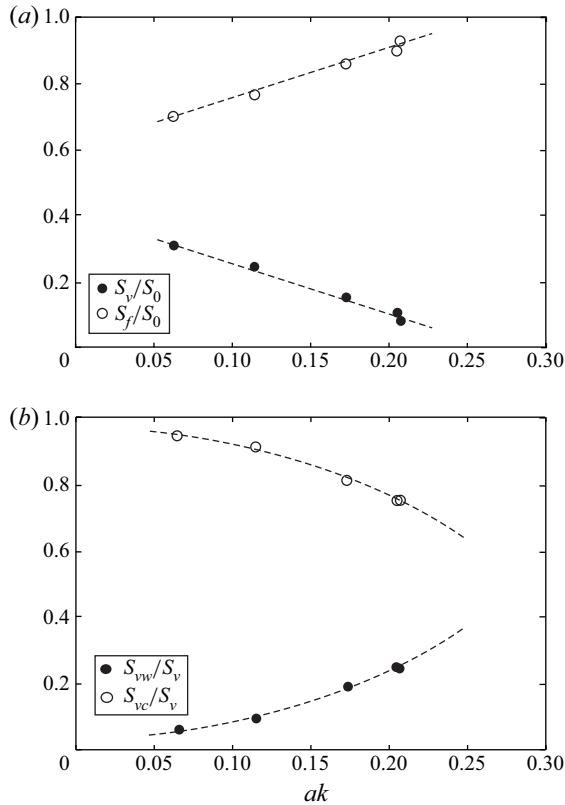


FIGURE 10. (a) Wind input terms from viscous forcing S_v and pressure forcing S_f normalized by the total wind input S_0 . (b) Normalized wind input terms from viscosity that contribute to currents S_{vc} and waves, S_{vw} .

wind input in these higher winds. It can be noted, however, that waves with $ak > 0.3$ are generally experiencing some form of breaking; this also generates surface currents (Rapp & Melville 1990; Melville, Veron & White 2002), albeit via a different pathway than through viscous stress.

Finally, the wind's energy input term to the wave field is (to the leading order)

$$\begin{aligned}
 S_{in} &= \overline{\tau_v \tilde{u}_s} + \overline{\tau_f c}, \\
 &= S_{vw} + S_f.
 \end{aligned}
 \tag{3.14}$$

Note that both viscous and pressure stresses act in a symmetrical fashion with respect to surface velocity: wave-phase-coherent viscous stress needs to correlate with \tilde{u}_s , and the pressure needs to correlate with \tilde{w}_s in order to contribute to wave growth. In other words, in order to contribute efficiently to wave growth, the viscous stress should be in phase with the water surface elevation. Likewise, the pressure should be in quadrature with the water surface elevation. In these conditions, wave-coherent pressure and viscous stresses with comparable amplitudes would contribute equally to wave growth (see also Longuet-Higgins 1969b). In this study, we find the ratio S_{vw}/S_{in} to be fairly constant regardless of wind and wave conditions; we find that the viscous tangential stress contributing to wave growth represents approximately $(3 \pm 0.5) \%$ of the total wind energy input. Here, two factors cause low relative contributions of viscous stress: First, the along-wave viscous stresses are not exactly in phase with the horizontal surface velocity,

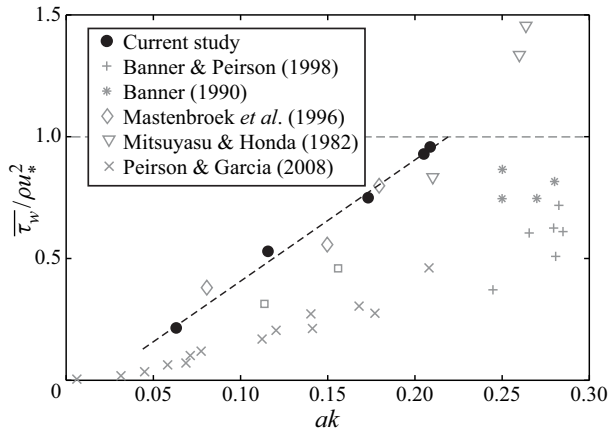


FIGURE 11. Fraction of wind stress contributing to the energy flux into the wave field: wave-coherent surface stress τ_w normalized by the total stress shown alongside the measurements of Mitsuyasu & Honda (1982), Banner (1990), Banner & Peirson (1998), Mastenbroek *et al.* (1996) and Peirson & Garcia (2008).

especially at low wind speeds. The phase shift does decrease with increasing wind speed (and wave slope), but then viscous forces just become negligible as a whole. Second, the amplitude of the along-wave viscous stresses is likely to be $O(0.1)$ of that of the along-wave pressure variations.

When normalized by the wave speed, the wind energy input (from pressure and viscous forces combined) can be expressed as a wave-coherent surface stress τ_w (e.g. Peirson & Garcia 2008; Grare *et al.* 2013a; Melville & Fedorov 2015)

$$\tau_w = \frac{S_{in}}{c}. \tag{3.15}$$

Figure 11 shows the ratio $\tau_w / \rho u_*^2$ which essentially shows the fraction of the total wind stress contributing to the energy flux into the wave field. Our results are plotted alongside other available measurements. They show a reasonable agreement with previous results albeit on the higher end of the range. This is likely because these laboratory waves are young and strongly forced. For small wave slopes ($ak < 0.1$), the wind imparts less than 50% of its momentum to the wave field. With increasing wave slope ak , steeper waves may induce greater modulations of the surface wind stress, thus yielding an increase in the fraction of surface wind stress (form stress plus viscous stress) contributing to the development of the wave field. Our measurements show that for waves with slopes ak greater than 0.2, the near entirety of the wind stress transfers momentum directly into wave motions.

4. Discussion

4.1. Wave growth

Neglecting local nonlinear wave–wave interactions and assuming horizontal homogeneity, (3.8) simply indicates that

$$\frac{\partial E_w}{\partial t} = S_{in} - D, \tag{4.1}$$

where $E_w = \frac{1}{2}\rho_w g a^2$ is the wave energy density of the peak dominant wave. Thus, the dimensionless wave growth rate is

$$\begin{aligned}\beta &= \frac{2\pi}{\omega} \frac{1}{E_w} \frac{\partial E_w}{\partial t} \\ &= \frac{2\pi}{\omega} \frac{1}{E_w} (S_{in} - D),\end{aligned}\quad (4.2)$$

in which $\omega = 2\pi f$ is the peak angular frequency and the wind input is $S_{in} = \tau_w c$ (3.15). Assuming that the bulk of the waves dissipation arises from water-side viscosity (Lamb 1932; Phillips 1977),

$$D = 2\nu_w \rho_w k^3 a^2 c^2, \quad (4.3)$$

with ν_w the water kinematic viscosity. Hence, the dimensionless wave growth rate can be written as

$$\begin{aligned}\beta &= \frac{2\pi}{\omega} \frac{2\tau_w c}{\rho_w g a^2} - \frac{2\pi}{\omega} \frac{4\nu_w \rho_w k^3 a^2 c^2}{\rho_w g a^2}, \\ &= \frac{4\pi \tau_w c}{\rho_w g a^2 k c} - \frac{8\pi \nu_w k^3 c^2}{g k c}, \\ &= \frac{4\pi \tau_w}{\rho_w c^2 a^2 k^2} - \frac{8\pi \nu_w k}{c}, \\ &= 4\pi \frac{\rho}{\rho_w} \frac{\tau_w}{\rho u_*^2} \frac{u_*^2}{c^2} \frac{1}{(ak)^2} - 8\pi Re^{-1},\end{aligned}\quad (4.4)$$

where we have used the dispersion relationship for linear gravity waves $\omega^2 = gk$, and where $Re = c/\nu_w k$ is the wave Reynolds number. As pointed out by Melville & Fedorov (2015), (4.4) shows that, dissipation aside, for a given wave slope and wave age, the problem of determining the wave growth rate essentially resides with the determination of the ratio of the wave-coherent momentum flux to the total momentum flux $\tau_w/\rho u_*^2$. Alternatively, it is also clear that wave slope, wave age and wave-coherent momentum flux are all essential ingredients to the wave growth process.

Figure 12 shows the dimensionless wave growth rate as a function of the inverse wave age. The open and closed black symbols and black crosses are data collected by Plant (1982). The coloured symbols are the present data and the solid line is Miles' theory (Miles 1959, 1993). Our data clearly fall within the scatter of the available historical dataset. They are also consistent with the numerical predictions of Yang, Meneveau & Shen (2013) and Melville & Fedorov (2015). The agreement is somewhat surprising for the cases with the highest wave slope for which we expect some small-scale microbreaking to take place and contribute to the dissipation, adding to the viscous effects.

From the first term in (4.4), we now isolate the dimensionless wave growth rate due to wind input alone

$$\beta_{in} = 4\pi \frac{\rho}{\rho_w} \frac{\tau_w}{\rho u_*^2} \frac{u_*^2}{c^2} \frac{1}{(ak)^2}, \quad (4.5)$$

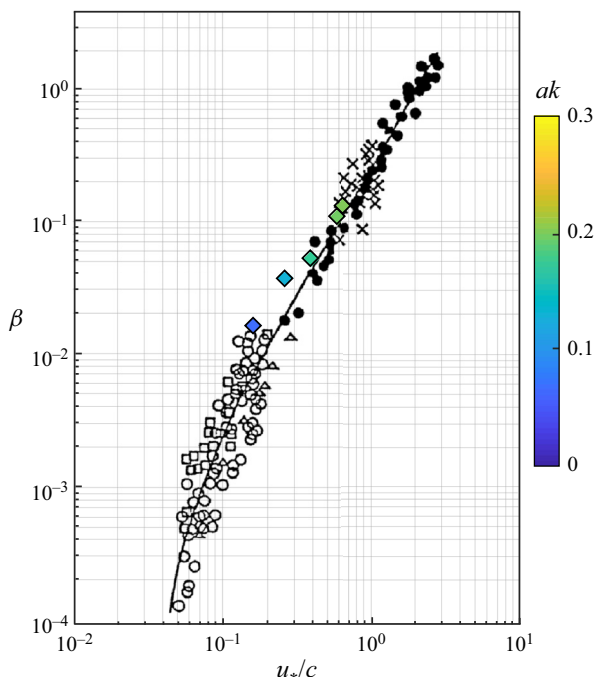


FIGURE 12. Dimensionless wave growth rate as a function of the inverse wave age. The open and closed black symbols and black crosses are data collected by Plant (1982) and plotted alongside Miles' theory (Miles 1959, 1993) (line) for a Charnock constant of $\alpha_{CH} = 0.01444$ (see Komen *et al.* 1994).

and further scale it with the wave age and the air–water density ratio

$$\begin{aligned}\beta_o &= \frac{1}{2\pi} \beta_{in} \frac{\rho_w c^2}{\rho u_*^2}, \\ &= 2 \frac{\tau_w}{(\rho u_*^2)} \frac{1}{(ak)^2}.\end{aligned}\quad (4.6)$$

Figure 13 shows estimates of β_o based on our estimates of τ_w . The solid black line ($\beta_o = 2(ak)^{-2}$) indicates the theoretical limit where the entire wind energy input would solely contribute to the wave field ($\tau_w = \rho u_*^2$ in (4.6)). When ak approaches 0.2, this limit is nearly reached. At smaller ak , the growth rate coefficient β_o departs dramatically from this limit, pointing to the reduced role of the wave field for the air–water energy flux at lower ak (see also figure 11). Our measurements of β_o agree remarkably well with those of Grare *et al.* (2013*b*) and Mastenbroek *et al.* (1996) but lie substantially above the rest of the available data.

4.2. Airflow separation effects

In figure 14, we report both the fraction of viscous stress over the total stress and the fraction of waves that experience airflow separation (noted α), as a function of three different dimensionless numbers: wave slope ak (figure 14*a*), wave age c/u_* (figure 14*b*) and roughness Reynolds number $z_0 u_* / \nu$ (figure 14*c*), for all our wind wave experiments.

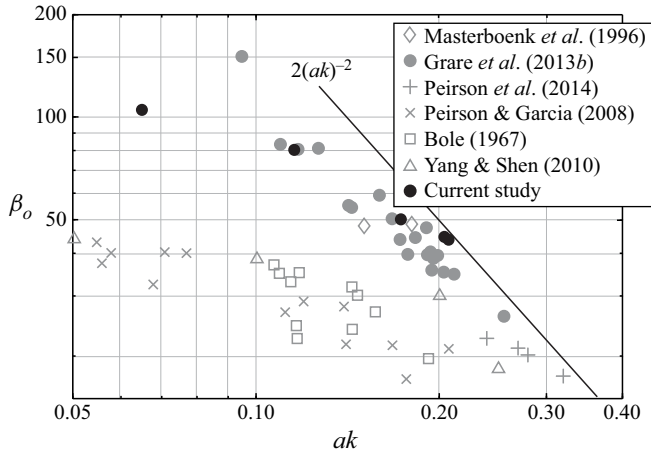


FIGURE 13. Wave growth coefficient β_o (see (4.6) for a definition of β_o). Our measurements (black filled circles) are presented alongside results of Mastenbroek *et al.* (1996), Grare *et al.* (2013a), Peirson & Garcia (2008), Peirson *et al.* (2014), Bole & Hsu (1969) and Yang & Shen (2010). The black line is the theoretical limit $\beta_o = 2(ak)^{-2}$ which assumes that $\tau_w = \rho u_*^2$ (see (4.6)).

In all cases, the normalized viscous stress decays from left to right, while the fraction of airflow separating waves increases. In this analysis, a wave is considered to experience airflow separation, when the instantaneous near-surface viscous stress drops below zero on the downwind face of the wave. At the lowest wind speed ($ak = 0.06$), the airflow separates over 0.4 % of the waves and the surface viscous stress represents 84 % of the total stress, whereas at the highest wind speed ($ak = 0.21$), 87 % of the waves experience airflow separation and the viscous stress is only 6.5 % of the total stress.

The influence of airflow separation is estimated by conditionally phase averaging the viscous stress and separating the waves that exhibit airflow separation from those that do not. The results are plotted in figures 15(b) and 15(c). The phase-averaged viscous stress across all waves, whether airflow separating or not, is repeated in panel (a), for comparison purposes (figure 7a). The results clearly show that airflow separation influences on average predominantly the stress at the crests and at the troughs: at crests, airflow separating waves experience higher than average crest viscous stress, while over troughs, they experience lower than average viscous stress. In other words, the crest-to-trough difference in surface viscous stress is substantially higher in waves that experience airflow separation, compared to the waves that do not. Also, the phase-averaged viscous stress at the lowest wind speed was removed from figure 15(c) because the sample size was too small, since at that low wind speed ($U_{10} = 2.19 \text{ m s}^{-1}$) only nine waves experienced airflow separation (out of the 2520 waves sampled at that wind speed).

In figure 16, we map the fractional number of waves inducing airflow separation, i.e. α , the number of airflow separating waves over the total number of waves in the $(\gamma, u_*/c_m)$ parameter space, where γ is the measured local wave steepness. Here, one $(\gamma, u_*/c_m)$ pair was estimated for each individual wave: c_m is the wave crest speed measured locally using 2 consecutive large field of view LIF wave profiles, and $\gamma = \pi H/\lambda$, with H the crest-to-trough wave height and λ the wavelength, also measured on instantaneous LIF wave profiles.

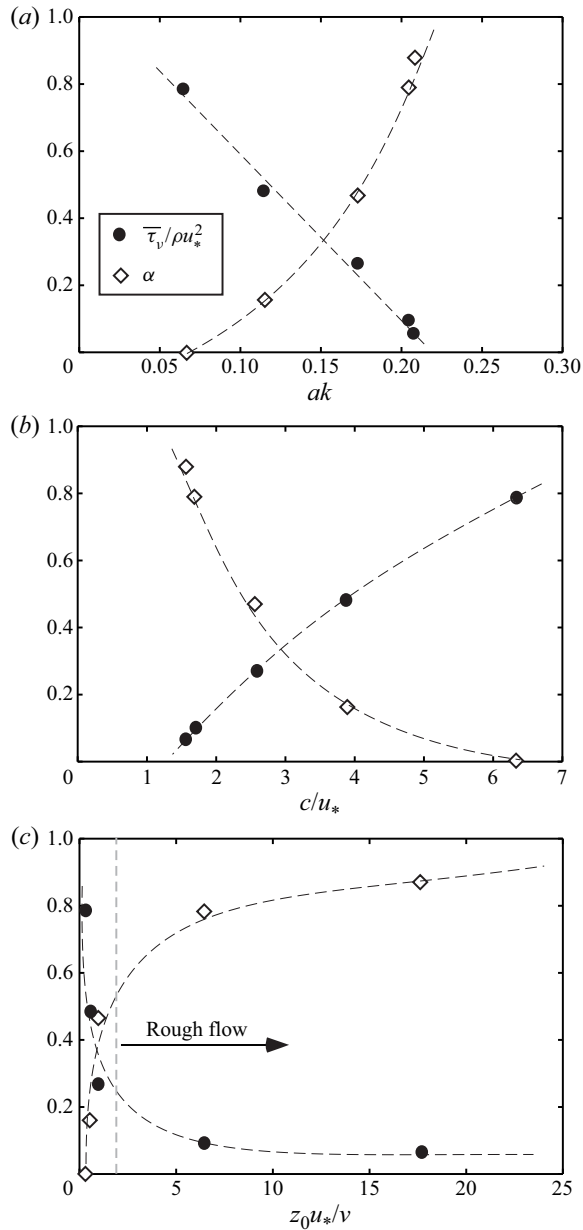


FIGURE 14. Fraction of viscous stress normalized by total stress, alongside fraction of airflow separating waves over total number of sampled waves, α , as a function of wave slope ak (a), wave age c/u_* (b) and roughness Reynolds number $z_0 u_* / \nu$ (c).

Since airflow separation is potentially tied to breaking (Banner & Melville 1976; Gent & Taylor 1977), it is interesting to compare these data with criteria for the onset of wave breaking. To that end, we use the kinematic criterion proposed by Banner & Phillips (1974) who suggest that wind waves may begin to break when

$$\frac{u_0}{c} = 1 - \sqrt{\frac{u_s}{c} \left(2 - \frac{u_s}{c} \right)}, \tag{4.7}$$

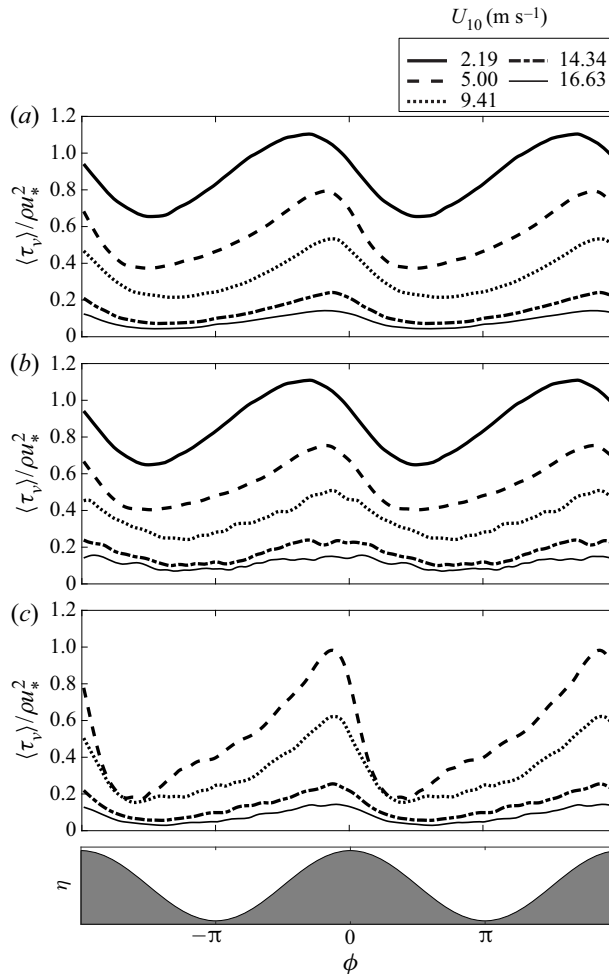


FIGURE 15. Phase-averaged viscous stress $\langle \tau_v \rangle$ for all waves (a), for only non-airflow separating waves (b) and for only airflow separating waves (c). A sketch of a mean wave profile is provided (bottom) to help visualize the wave phase.

where u_0 is the maximum horizontal orbital velocity at the water surface, and u_s the surface drift. Here, we estimate the surface drift as $u_s = 0.55u_*$ (Phillips & Banner 1974). Our estimate from this dataset suggests that $u_s = 0.9u_*$, but this difference has no consequence on this discussion. The ratio u_0/c in (4.7) is replaced by the wave steepness γ (since $u_0 \propto akc$ in the linear approximation, and since $ak \sim \gamma$). We note that the limit suggested by Banner & Phillips (1974) (dashed curve) is in the vicinity of the 50% contour line (solid line) which denotes where half of the waves are experiencing/ causing airflow separation. However, (4.7) departs from the 50% occurrence of separation in more extreme wind forcing conditions (higher inverse wave ages) where the apparent asymptotic behaviour suggests that wave slope controls airflow separation in those regimes, regardless of wave age. In the strongly forced conditions where waves propagate much slower than the wind, it is reasonable to think that, as aerodynamic flows over a fixed obstacle, it is the geometrical properties of the surface which control the occurrence of flow separation. Overall, we find

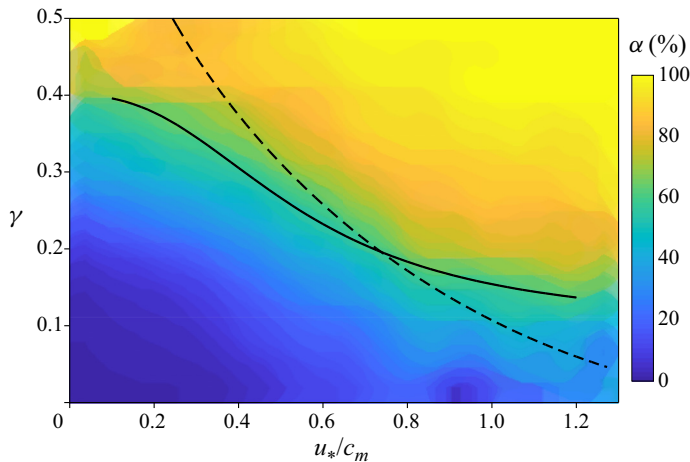


FIGURE 16. Relative fraction α of airflow separating waves over total number of waves, as a function of wave slope and inverse wave age. The dashed line represents the incipient breaking limit suggested by Banner & Phillips (1974) (4.7). The solid line is the fit to the 50% contour line given by (4.8).

that

$$\gamma = 0.1 + \frac{0.3}{1 + \left(\frac{u_*/c_m}{0.3}\right)^{5/2}} \quad (4.8)$$

describes well the 50% contour line. For example, in low winds with $u_*/c_m \lesssim 0.1$, airflow separates if the wave steepness approaches 0.4. We note that in the absence of wind, the Stokes' limit for periodic waves of highest slope gives $ak = 0.4432$. In higher winds where $u_*/c_m \gtrsim 0.9$, the airflow will be predominantly separated for waves with $\gamma > 0.1$.

5. Conclusions

In this paper, we have presented high resolution velocity measurements in the airflow above laboratory wind waves within and above the airflow's viscous sublayer, for a range of different wind-wave conditions (winds ranging from 2.19 to 16.63 m s⁻¹) and at a fetch of 22.7 m. The high quality and resolution of the wind and wave profile measurements provided us with surface stress estimates along the air-water interface of thousands of waves for each wind-wave condition. We were also able to systematically detect the occurrence of airflow separation past wave crests, using a robust surface viscous stress criterion.

The airflow intermittently separates past wave crests, starting at wind speeds as low as 2.2 m s⁻¹. Such airflow separation events cause the surface viscous stress to peak in the vicinity of the wave's crest, and to dramatically drop to below zero values past the wave's crest. When the airflow does not separate, sheltering remains important, and the surface stress decreases more gradually past wave crests. Airflow separation is rare over low wind speed, small slope wind waves, but occurs with increasing frequency when wind speed and thereby wave slopes increase. At the highest measured wind speed ($U_{10} = 16.63$ m s⁻¹), nearly 100% of the waves experience airflow separation past their crests. In these conditions, it is likely a large percentage of waves are also (micro) breaking.

This suggests that breaking may be a sufficient (but not necessary) kinematic condition leading to systematic airflow separation (Banner & Melville 1976). However, separation is also observed in benign conditions, albeit intermittently. Our results suggest that in low winds, geometric properties of the wave such as wave slope and/or wave asymmetry, rather than kinematic conditions, lead to airflow separation.

Finally, we were able to quantify the partitioning of the total surface momentum and energy fluxes across the air–water interface, between surface normal pressure and surface tangential viscous contributions, which predominantly contribute to wave growth and to surface wind drift currents, respectively. Indeed in this study, in spite of strong wave-phase locked features, (e.g. maximum near wave crests, and minimum near wave troughs), viscous stress does not significantly contribute to wave growth. While wave growth is the main pathway for air–water momentum and mechanical energy fluxes at high wave slopes, the development of surface currents (wind drift) under the action of tangential viscous shear is an important flux mechanism when wave slopes are small.

Acknowledgements

This work was supported by US National Science Foundation grants OCE-1458977, OCE-1233808, OCE 0748767 and AGS-PRF-1524733. M.P.B. acknowledges support from the Deutsche Forschungsgemeinschaft (DFG, German Research Foundation, project number 274762653, Collaborative Research Centre TRR 181 ‘Energy Transfers in Atmosphere and Ocean’). M.P.B. thanks Jeff Carpenter for fruitful discussions on wind–wave interactions.

Declaration of interests

The authors report no conflict of interest.

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